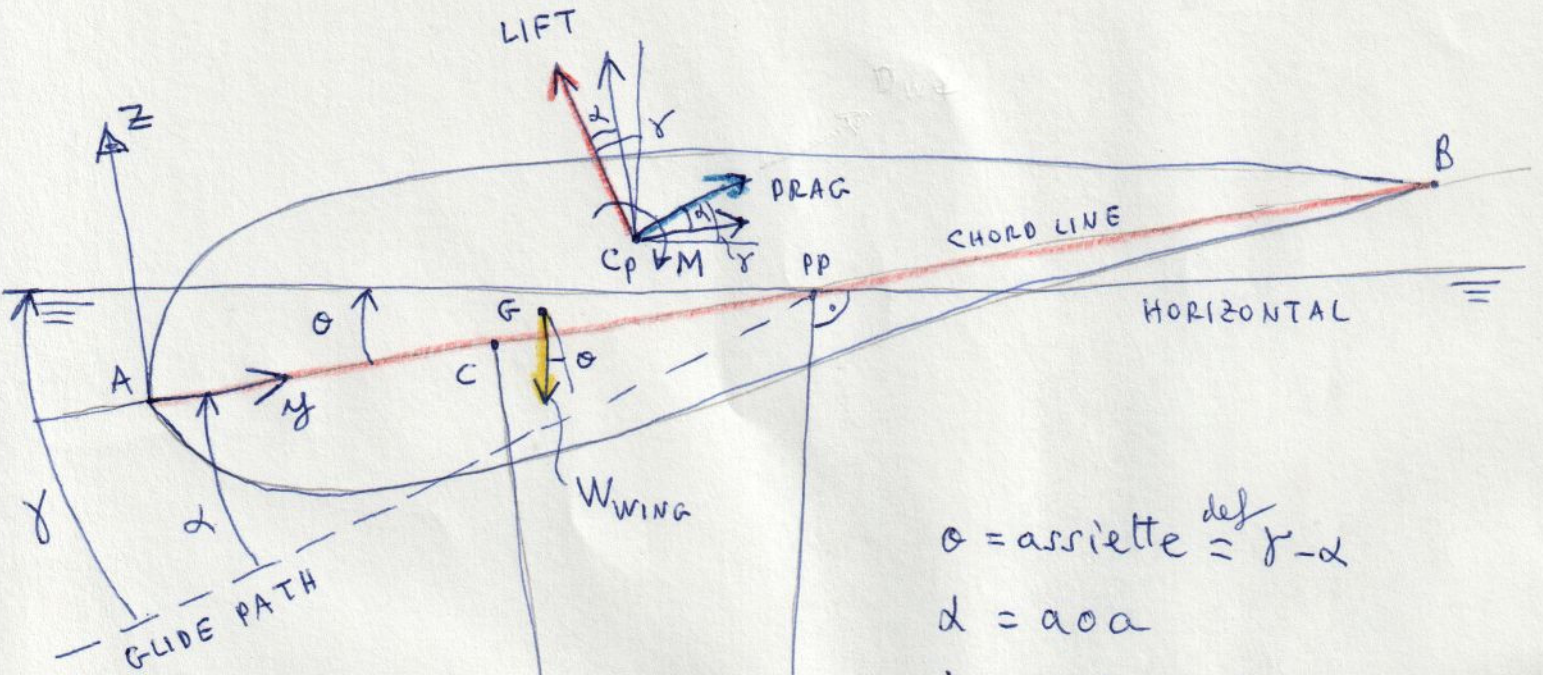


PARAGLIDER LONGITUDINAL EQUILIBRIUM



$$\theta = \text{assiette} \stackrel{\text{def}}{=} \gamma - \alpha$$

$$\alpha = \text{aoa}$$

$$\gamma = \text{glide angle}$$

$$GR = \frac{1}{\tan(\gamma)}$$

\overline{AB} chord line

\overline{AC} calage

G wing cog

C_p center of pressure

PP Plumb point

P pilot position

$\overline{PC} = h$ lines height

DL \rightarrow DRAG LINES

GL

LINES

VERTICAL

EQUILIBRIUM

$$\sum H = 0$$

$$\sum V = 0$$

$$\sum M = 0$$

